

Note 361(1): General Dynamics with the Lagrange Derivative

General dynamics is developed with the Euler equation:

$$\underline{a} = \frac{D\underline{v}}{Dt} = \frac{d\underline{v}}{dt} + (\underline{v} \cdot \nabla) \underline{v} \quad - (1)$$

where \underline{a} is the acceleration. As shown in UFT 351, eq. (1) is an example of the Cartan covariant derivative:

$$\frac{D\underline{v}^a}{dx^u} = \frac{d\underline{v}^a}{dx^u} + \omega^a_{ub} \underline{v}^b \quad - (2)$$

where \underline{v}^a is a vector in any dimension and ω^a_{ub} is the spin connection.

Therefore general dynamics is developed automatically into a theory of general relativity.

In Cartesian coordinates eq. (1) is three dimensional

$$\frac{D\underline{v}}{Dt} = \frac{d\underline{v}}{dt} + \left(v_x \frac{d}{dx} + v_y \frac{d}{dy} + v_z \frac{d}{dz} \right) \underline{v} \quad - (3)$$

$$\frac{Dv_x}{Dt} = \frac{dv_x}{dt} + v_x \frac{dv_x}{dx} + v_y \frac{dv_x}{dy} + v_z \frac{dv_x}{dz} \quad - (4)$$

$$\frac{Dv_y}{Dt} = \frac{dv_y}{dt} + v_x \frac{dv_y}{dx} + v_y \frac{dv_y}{dy} + v_z \frac{dv_y}{dz} \quad - (5)$$

$$\frac{Dv_z}{Dt} = \frac{dv_z}{dt} + v_x \frac{dv_z}{dx} + v_y \frac{dv_z}{dy} + v_z \frac{dv_z}{dz} \quad - (6)$$

If we define:

$$\underline{v} = \begin{bmatrix} v_x \\ v_y \\ v_z \end{bmatrix} \quad \text{--- (7)}$$

--- (8)

then: $\frac{D}{Dt} \begin{bmatrix} v_x \\ v_y \\ v_z \end{bmatrix} = \frac{d}{dt} \begin{bmatrix} v_x \\ v_y \\ v_z \end{bmatrix} + \begin{bmatrix} \partial v_x / \partial x & \partial v_x / \partial y & \partial v_x / \partial z \\ \partial v_y / \partial x & \partial v_y / \partial y & \partial v_y / \partial z \\ \partial v_z / \partial x & \partial v_z / \partial y & \partial v_z / \partial z \end{bmatrix} \begin{bmatrix} v_x \\ v_y \\ v_z \end{bmatrix}$

where the matrix is the Jacobian. It defines the spiral conversion matrix:

$$\begin{bmatrix} \omega^1_{01} & \omega^1_{02} & \omega^1_{03} \\ \omega^2_{01} & \omega^2_{02} & \omega^2_{03} \\ \omega^3_{01} & \omega^3_{02} & \omega^3_{03} \end{bmatrix} = \begin{bmatrix} \partial v_x / \partial x & \partial v_x / \partial y & \partial v_x / \partial z \\ \partial v_y / \partial x & \partial v_y / \partial y & \partial v_y / \partial z \\ \partial v_z / \partial x & \partial v_z / \partial y & \partial v_z / \partial z \end{bmatrix} \quad \text{--- (9)}$$

From the ECE2 unification of gravitation and fluid dynamics it follows that the Newtonian acceleration due to gravity:

$$\underline{g} = \frac{d\underline{v}}{dt} = -\underline{\nabla} \phi = -\frac{m M G}{r^2} \quad \text{--- (10)}$$

generalized to:

$$\underline{g} = \frac{d\underline{v}}{dt} + (\underline{v} \cdot \underline{\nabla}) \underline{v} = -\underline{\nabla} \phi \quad \text{--- (11)}$$

i.e. to an Euler equation.

For a planar orbit it follows that:

$$\frac{Dv_x}{Dt} = \frac{\partial v_x}{\partial t} + v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} \quad - (12)$$

$$\frac{Dv_y}{Dt} = \frac{\partial v_y}{\partial t} + v_x \frac{\partial v_y}{\partial x} + v_y \frac{\partial v_y}{\partial y} \quad - (13)$$

The non-Newtonian part of the acceleration \underline{g} in eq. (11) is:

$$\underline{g}_1 = (\underline{v} \cdot \underline{\nabla}) \underline{v} \quad - (14)$$

and the Newtonian or vertical part of the acceleration

$$\underline{g}_N = \frac{d\underline{v}}{dt}, \quad - (15)$$

so the complete acceleration is:

$$\underline{g} = \underline{g}_N + \underline{g}_1 \quad - (16)$$

In immediately preceding pages it has been shown that the velocity:

$$\underline{v} = \frac{(MG)^{1/2} (-X\underline{i} + Y\underline{j})}{(X^2 + Y^2)^{3/4}} \quad - (16)$$

used in eq. (14) produces:

$$\underline{g}_1 = -\frac{MG}{(X^2 + Y^2)^{1/2}} \underline{e}_r \quad - (17)$$

and \underline{v} was identified with the orbital velocity in a plane.

7) In ECE2 / Kambe fluid dynamics:

$$\underline{g}_1 = (\underline{v} \cdot \underline{\nabla}) \underline{v} = -\frac{\partial \underline{v}}{\partial t} - \underline{\nabla} \phi \quad (18)$$

where \underline{v} is the orbital velocity. From eq. (16) it follows that:

$$v_x = -\frac{(mG)^{1/2} x}{(x^2 + y^2)^{3/4}} \quad (19)$$

$$v_y = \frac{(mG)^{1/2} y}{(x^2 + y^2)^{3/4}} \quad (20)$$

It is clear that the orbital velocity produced by the non-Newtonian part of the acceleration, and also, the Lagrange derivative of the orbital velocity. The Newtonian acceleration between n and M would result if n was attracted to M in a line joining n and M . The Newtonian acceleration alone does not produce an orbit.

For a circular orbit:

$$x^2 + y^2 = r^2 = d^2 \quad (21)$$

where d is the half right latitude. ^{constant} So for a circular orbit:

$$\frac{\partial v_x}{\partial x} = -\left(\frac{mG}{d^3}\right)^{1/2}, \quad \frac{\partial v_x}{\partial y} = 0, \quad (22)$$

$$\frac{\partial v_y}{\partial x} = 0, \quad \frac{\partial v_y}{\partial y} = \left(\frac{mG}{d^3}\right)^{1/2} \quad (23)$$

5) So the Jacobian matrix of spacetime is:

$$\begin{bmatrix} \omega^{1.01} & \omega^{1.02} \\ \omega^{2.01} & \omega^{2.02} \end{bmatrix} = \left(\frac{MG}{d^3} \right)^{1/2} \begin{bmatrix} -1 & 0 \\ 0 & 1 \end{bmatrix} \quad (24)$$

The presence of a spacetime means that the theory is a theory of general relativity based on Cartesian geometry. Without the spacetime there is no orbit.

For a circular orbit the non-Newtonian acceleration is:

$$\underline{g}_1 = -\frac{mMG}{r^2} \underline{e}_r = (\underline{v} \cdot \underline{\nabla}) \underline{v} \quad (25)$$

where

$$\underline{v}^2 = \frac{MG}{r} \quad (26)$$

"the square of the orbital velocity". For a circular orbit this is constant:

$$\underline{v}^2 = \frac{MG}{d} \quad (27)$$

From the theory of fluid gravitation (ECC2/Kantle fluid dynamics):

$$\underline{g}_1 = -\frac{mMG}{r^2} \underline{e}_r = -\frac{\partial \underline{v}}{\partial t} - \underline{\nabla} \phi \quad (28)$$

where ϕ is the gravitational potential:

$$\phi = -\frac{MG}{r} \quad (29)$$

It follows that for a circular orbit:

$$6) \quad \frac{\partial \underline{v}}{\partial t} = \frac{\partial^2 \underline{r}}{\partial t^2} = 0 \quad - (30)$$

which is consistent with eq. (27).

Using the result:

$$\alpha = \frac{L^2}{m^2 m b} \quad - (31)$$

where L is the constant angular momentum of the system
it follows that:

$$\begin{bmatrix} \omega^1_{01} & \omega^1_{02} \\ \omega^2_{01} & \omega^2_{02} \end{bmatrix} = \left(\frac{m}{L}\right)^3 (mb)^2 \begin{bmatrix} -1 & 0 \\ 0 & 1 \end{bmatrix} \quad - (32)$$

Finally, the general force equation for all orbits

is:

$$\underline{F} = m \left(\frac{\partial \underline{v}}{\partial t} + (\underline{v} \cdot \underline{\nabla}) \underline{v} \right) = -m \underline{\nabla} \phi \quad - (33)$$