

### 363(1): Orbital Precession due to the Fluid Vacuum

As shown in Note 362(5), the orbital velocity in the presence of a fluid spacetime, after or vacuum is given in component form by:

$$\frac{D}{Dt} \begin{bmatrix} r \\ 0 \end{bmatrix} = \frac{d}{dt} \begin{bmatrix} r \\ 0 \end{bmatrix} + \begin{bmatrix} 0 & -\dot{\theta} \\ \dot{\theta} & 0 \end{bmatrix} \begin{bmatrix} r \\ 0 \end{bmatrix} + \begin{bmatrix} \Omega^{1}_{01} & \Omega^{1}_{02} \\ \Omega^{2}_{01} & \Omega^{2}_{02} \end{bmatrix} \begin{bmatrix} \dot{r} \\ r\dot{\theta} \end{bmatrix} - (1)$$

In the notation of note 362(5). In the absence of the fluid vacuum:

$$\frac{D}{Dt} \begin{bmatrix} r \\ 0 \end{bmatrix} = \frac{d}{dt} \begin{bmatrix} r \\ 0 \end{bmatrix} + \begin{bmatrix} 0 & -\dot{\theta} \\ \dot{\theta} & 0 \end{bmatrix} \begin{bmatrix} r \\ 0 \end{bmatrix} - (2)$$

It is shown in this note that eq. (1) produces orbital precession due to the fluid spacetime, after or vacuum. In eq. (1), the Cartan Spicometria matrix is:

defined by:

$$\Omega^{a}_{0b} = \begin{bmatrix} \Omega^{1}_{01} & \Omega^{1}_{02} \\ \Omega^{2}_{01} & \Omega^{2}_{02} \end{bmatrix} = \begin{bmatrix} \frac{\partial r_r}{\partial r} & \frac{1}{r} \frac{\partial r_r}{\partial \theta} \\ \frac{\partial r_\theta}{\partial r} & \frac{1}{r} \frac{\partial r_\theta}{\partial \theta} \end{bmatrix} - (3)$$

Eq. (2) gives the Coriolis velocity components:

$$V_r = \dot{r} - (4)$$

$$V_\theta = \dot{\theta} r = \omega r, - (5)$$

this familiar result can be thought of as the result in the absence of a fluid vacuum.

Eq. (1) gives the effect on the orbit of a fluid vacuum:

$$2) \quad v_r = \dot{r} + \Omega^1_{01} \dot{r} + \Omega^1_{02} r \dot{\theta} \quad - (6)$$

$$v_\theta = \omega r + \Omega^2_{01} \dot{r} + \Omega^2_{02} r \dot{\theta} \quad - (7)$$

Therefore:

$$v_r = (1 + \Omega^1_{01}) \dot{r} + \Omega^1_{02} \omega r \quad - (8)$$

$$v_\theta = (1 + \Omega^2_{02}) \omega r + \Omega^2_{01} \dot{r} \quad - (9)$$

For a Newtonian orbit:

$$v_r = \dot{r} = \frac{dr}{dt} \quad - (10)$$

and

$$v_\theta = \omega r \quad - (11)$$

Use:

$$\frac{dr}{dt} = \frac{dr}{d\theta} \frac{d\theta}{dt} = \frac{L}{mr^2} \frac{dr}{d\theta} \quad - (12)$$

and

$$v_\theta = r \frac{d\theta}{dt} \quad - (13)$$

It follows that:

$$v_r = \frac{d\theta}{dt} \frac{dr}{d\theta} \quad - (14)$$

and that:

$$\frac{v_r}{v_\theta} = \frac{1}{r} \frac{dr}{d\theta} \quad - (15)$$

Now denote  $v_r$  and  $v_\theta$  of a precessing orbit by  $v_r(P)$  and  $v_\theta(P)$  and those of a Newtonian orbit by  $v_r(N)$  and  $v_\theta(N)$ . Then eqs. (8) and (9) become:

$$v_r(P) = (1 + \Omega^1_{01}) v_r(N) + \Omega^1_{02} v_\theta(N) \quad - (16)$$

$$v_\theta(P) = (1 + \Omega^2_{02}) v_\theta(N) + \Omega^2_{01} v_r(N) \quad - (17)$$

From eq. (15):

$$\frac{V_r(N)}{V_\theta(N)} = \frac{1}{r} \frac{dr}{d\theta} \quad - (18)$$

where

$$r = \frac{\alpha}{1 + \epsilon \cos \theta} \quad - (19)$$

so

$$\frac{dr}{d\theta} = \frac{\epsilon r^2 \sin \theta}{\alpha} \quad - (20)$$

so

$$\frac{V_r(N)}{V_\theta(N)} = \frac{\epsilon r \sin \theta}{\alpha} \quad - (21)$$

in which

$$\sin \theta = \left( 1 - \cos^2 \theta \right)^{1/2} \quad - (22)$$

$$= \left( 1 - \frac{1}{\epsilon^2} \left( \frac{\alpha}{r} - 1 \right)^2 \right)^{1/2}$$

so

$$\frac{V_r(N)}{V_\theta(N)} = \frac{\epsilon r}{\alpha} \left( 1 - \frac{1}{\epsilon^2} \left( \frac{\alpha}{r} - 1 \right)^2 \right)^{1/2} \quad - (23)$$

and

$$\frac{V_\theta(N)}{V_r(N)} = \frac{\alpha}{\epsilon r} \left( 1 - \frac{1}{\epsilon^2} \left( \frac{\alpha}{r} - 1 \right)^2 \right)^{-1/2} \quad - (24)$$

If we denote:

$$A = \frac{\epsilon r}{\alpha} \left( 1 - \frac{1}{\epsilon^2} \left( \frac{\alpha}{r} - 1 \right)^2 \right)^{1/2} \quad - (25)$$

then eqs (16) and (17) become relations  
between  $V_r(P)$  and  $V_\theta(P)$  and  $V_r(N)$  and  $V_\theta(N)$ :

$$\therefore V_R(\underline{P}) = \left( 1 + \Omega^1_{01} + \frac{\Omega^1_{02}}{A} \right) V_R(N) - (28)$$

$$\text{and } V_\theta(\underline{P}) = (1 + \Omega^2_{02} + A\Omega^2_{01}) V_\theta(N) - (29)$$

It follows that:

$$\frac{V_R(\underline{P})}{V_\theta(\underline{P})} = B \frac{V_R(N)}{V_\theta(N)} - (28)$$

where:

$$B = \frac{1 + \Omega^1_{01} + \frac{\Omega^1_{02}}{A}}{1 + \Omega^2_{02} + A\Omega^2_{01}} - (29)$$

so

$$\frac{V_R(\underline{P})}{V_\theta(\underline{P})} = AB - (30)$$

Now

we:

$$\begin{aligned} \underline{V}(\underline{P}) &= V_r(\underline{P}) \underline{e}_r + V_\theta(\underline{P}) \underline{e}_\theta - (31) \\ &= \dot{r}(\underline{P}) \underline{e}_r + \omega(\underline{P}) r(\underline{P}) \underline{e}_\theta \end{aligned}$$

then:

$$\frac{V_R(\underline{P})}{V_\theta(\underline{P})} = \left( \frac{dr}{d\theta} \right)_P - (32)$$

and

$$\boxed{\left( \frac{dr}{d\theta} \right)_P = B \left( \frac{dr}{d\theta} \right)_N} - (33)$$

1) If the processing unit is modelled by

$$r = \frac{d}{1 + \epsilon \cos(x\theta)} \quad - (34)$$

with

$$x = \frac{1 + 3\frac{MG}{c^2 d}}{c^2 d} \quad - (35)$$

then

$$\left(\frac{dr}{d\theta}\right)_P = \frac{x \epsilon r^2 \sin(x\theta)}{d} \quad - (36)$$

and

$$\left(\frac{dr}{d\theta}\right)_N = \frac{\epsilon r^2 \sin\theta}{d} \quad - (37)$$

To an excellent approximation:

$$\sin(x\theta) \sim \sin\theta \quad - (38)$$

in the solar system, where  $x$  is very close to unity.

Therefore

$$\left(\frac{dr}{d\theta}\right)_P = x \left(\frac{dr}{d\theta}\right)_N \quad - (39)$$

Comparing eqs. (33) and (39):

$$\boxed{x = 1} \quad - (40)$$

is the approximation (38).

From the experimental data it is known that  $x$  is the constant (35) for a given half right latitude  $d$ .

b) Therefore no possible solution is:

$$\frac{1 + \Omega^1_{01}}{1 + \Omega^2_{02}} = x = \frac{1 + \frac{3mG}{c^2}}{c^2} - (34)$$

and:  $\Omega^1_{02} = \Omega^2_{01} = 0 - (35)$

It has been shown that orbital precession to any precision is produced by a fluid spacetime, vacuum or aether.

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