

237(1) : Summary of Results and New Equation of Orbital Velocity in a plane

In general the acceleration associated with planar rotation is:

$$\underline{a} = \frac{d^2 \underline{r}}{dt^2} = \frac{d^2 r}{dt^2} \underline{e}_r + \underline{\omega} \times (\underline{\omega} \times \underline{r}) + \frac{d\underline{\omega}}{dt} \times \underline{r} + 2\underline{\omega} \times \frac{d\underline{r}}{dt} \quad - (1)$$

The Coriolis acceleration is:

$$\begin{aligned} \underline{a}_{\text{Coriolis}} &= \frac{d\underline{\omega}}{dt} \times \underline{r} + 2\underline{\omega} \times \frac{d\underline{r}}{dt} \\ &= \left(r \frac{d^2 \theta}{dt^2} + 2 \frac{dr}{dt} \frac{d\theta}{dt} \right) \underline{e}_\theta. \end{aligned} \quad - (2)$$

This is zero for any planar rotation because:

$$\underline{L} = \underline{r} \times \underline{p} = mr^2 \frac{d\theta}{dt} \underline{k} \quad - (3)$$

and $\frac{d\theta}{dt} = \frac{L}{mr^2} \quad - (4)$

So
$$\begin{aligned} \frac{d}{dt} \left(\frac{d\theta}{dt} \right) &= \frac{d}{dt} \left(\frac{L}{mr^2} \right) \\ &= \frac{dr}{dt} \frac{d}{dr} \left(\frac{L}{mr^2} \right) \quad - (4) \\ &= -\frac{2L}{mr^3} \frac{dr}{dt} \end{aligned}$$

So:

$$r \frac{d^2 \theta}{dt^2} + 2 \frac{dr}{dt} \frac{d\theta}{dt} = - \frac{2L}{mr^3} \frac{dr}{dt} + 2 \frac{dr}{dt} \frac{L}{mr^3}$$
$$= 0 \quad - (5)$$

(QED)

Therefore for any planar orbit:

$$\underline{a} = \frac{d^2 \underline{r}}{dt^2} = \underline{e}_r + \underline{\omega} \times (\underline{\omega} \times \underline{r}), \quad - (6)$$

In this equation:

$$f := \frac{dr}{dt} = \frac{dr}{d\theta} \frac{d\theta}{dt} = \frac{L}{mr^2} \frac{dr}{d\theta} \quad - (7)$$

and

$$\frac{df}{dt} = \frac{df}{dr} \frac{dr}{dt} = f \frac{df}{dr} \quad - (8)$$

$$\text{So: } \frac{d^2 \underline{r}}{dt^2} = \left(\frac{L}{mr} \right)^2 \left(\frac{dr}{d\theta} \right) \frac{d}{dr} \left(\frac{1}{r^2} \frac{dr}{d\theta} \right) \quad - (9)$$

Now note that:

$$\frac{d}{dr} \left(\frac{1}{r^2} \frac{dr}{d\theta} \right) = \frac{d\theta}{dr} \frac{d}{d\theta} \left(\frac{1}{r^2} \frac{dr}{d\theta} \right) \quad - (10)$$

$$\text{So } \frac{d^2 \underline{r}}{dt^2} = \left(\frac{L}{mr} \right)^2 \frac{d}{d\theta} \left(\frac{1}{r^2} \frac{dr}{d\theta} \right) \quad - (11)$$

Next note that:

$$3) \frac{d}{dt} \left(\frac{1}{r} \right) = \frac{d}{dr} \left(\frac{1}{r} \right) \frac{dr}{dt} = -\frac{1}{r^2} \frac{dr}{dt} \quad - (12)$$

$$\text{so } \frac{d^2 r}{dt^2} = - \left(\frac{L}{mr} \right)^2 \frac{d^2}{dt^2} \left(\frac{1}{r} \right) \quad - (13)$$

Therefore for any planar orbit:

$$\underline{a} = - \left(\frac{L}{mr} \right)^2 \frac{d^2}{dt^2} \left(\frac{1}{r} \right) \underline{e}_r + \underline{\omega} \times (\underline{\omega} \times \underline{r}) \quad - (14)$$

in which

$$\begin{aligned} \underline{\omega} \times (\underline{\omega} \times \underline{r}) &= -\omega^2 r \underline{e}_r \\ &= -\frac{L^2}{mr^3} \underline{e}_r \end{aligned} \quad - (15)$$

so:

$$\underline{a} = - \left(\frac{L}{mr} \right)^2 \left(\frac{d^2}{dt^2} \left(\frac{1}{r} \right) + \frac{1}{r} \right) \underline{e}_r \quad - (16)$$

The acceleration of any planar orbit is determined entirely by the spin constant:

$$\underline{\omega} = \omega \underline{k} = \frac{L}{mr^2} \underline{k} \quad - (17)$$

and \underline{a} is always directed along $-\underline{e}_r$. This is a result of pure kinematics.

4) At this point, force is defined as:

$$\underline{F} = m \underline{a} \quad - (18)$$

So

$$\underline{F} = - \frac{L^2}{mr^3} \left(\frac{d^2}{d\theta^2} \left(\frac{1}{r} \right) + \frac{1}{r} \right) \underline{e}_r \quad - (19)$$

for any planar orbit.

Lagrangian dynamics also give eq. (19), and are more general than Newtonian dynamics.

The orbital velocity is given by:

$$\underline{v} = \frac{dr}{dt} \underline{e}_r + \omega r \underline{e}_\theta \quad - (20)$$

$$= \frac{dr}{dt} \underline{e}_r + \underline{\omega} \times \underline{r}$$

From eq. (12):

$$\frac{dr}{d\theta} = -r^2 \frac{d}{d\theta} \left(\frac{1}{r} \right) = \frac{dr}{dt} \frac{dt}{d\theta} \quad - (21)$$

So

$$\frac{dr}{dt} = -\omega r^2 \frac{d}{d\theta} \left(\frac{1}{r} \right) \quad - (22)$$

It follows that:

$$\underline{v} = -\omega r^2 \frac{d}{d\theta} \left(\frac{1}{r} \right) \underline{e}_r + \omega r \underline{e}_\theta \quad - (23)$$

5) i.e. for any planar orbit:

$$\underline{v} = \frac{L}{mr} \underline{e}_\theta - \omega r^2 \frac{d}{d\theta} \left(\frac{1}{r} \right) \underline{e}_r \quad - (24)$$

$$\underline{v} = \left(\frac{L}{m} \right) \left(\frac{1}{r} \underline{e}_\theta - \frac{d}{d\theta} \left(\frac{1}{r} \right) \underline{e}_r \right) \quad - (25)$$

It follows that:

$$v^2 = \left(\frac{L}{m} \right)^2 \left[\frac{1}{r^2} + \left(\frac{d}{d\theta} \left(\frac{1}{r} \right) \right)^2 \right] \quad - (26)$$

for any planar orbit:

The kinetic energy of any planar orbit is:

$$T = \frac{1}{2} m v^2 = \frac{L^2}{2m} \left(\frac{1}{r^2} + \left(\frac{d}{d\theta} \left(\frac{1}{r} \right) \right)^2 \right) \quad - (27)$$

The potential energy of any planar orbit is

given by: $F = - \frac{\partial U}{\partial r} \quad - (28)$

from eqs. (19) and (28):

$$U = \int \frac{L^2}{mr^3} \left(\frac{d^2}{d\theta^2} \left(\frac{1}{r} \right) + \frac{1}{r} \right) dr \quad - (29)$$

6) The Hamiltonian is :

$$H = E = T + U \quad - (30)$$

and Lagrangian is :

$$L = T - U \quad - (31)$$

The limiting velocity is :

$$\underline{v} \xrightarrow{r \rightarrow \infty} - \frac{L}{m} \frac{d}{dt} \left(\frac{1}{r} \right) \underline{e}_r \quad - (32)$$

or

$$v = v_{\infty} \xrightarrow{r \rightarrow \infty} - \frac{L}{m} \frac{d}{dt} \left(\frac{1}{r} \right) \quad - (33)$$

i.e

$$\boxed{\frac{d}{dt} \left(\frac{1}{r} \right) = - \frac{mv_{\infty}}{L} \quad \text{constant} \quad - (34)}$$

for any planar orbit.

For a whirlpool galaxy :

$$v_{\infty} = \text{constant} \neq 0 \quad - (35)$$

so

$$\boxed{\frac{d}{dt} \left(\frac{1}{r} \right) = - \frac{1}{r_0}} \quad - (36)$$

where

$$r_0 = \frac{L}{mv_{\infty}} \quad - (37)$$

Therefore:

$$\boxed{r = -\frac{r_0}{\theta}} \quad (38)$$

and the orbit is a hyperbolic spiral as shown (QED)

From eqs. (19) and (38):

$$\boxed{F = -\frac{L^2}{mr^3} \underline{e}_r} \quad (39)$$

and the force law is inverse cube in r.

Eq. (39) means that the force needed to keep a star in a hyperbolic spiral orbit is inverse cube in r and directed towards the centre of the galaxy.

In Newtonian dynamics:

$$\frac{1}{r} = \frac{1}{d} (1 + e \cos \theta) \quad (40)$$

so
$$\frac{d}{d\theta} \left(\frac{1}{r} \right) = -\frac{e}{d} \sin \theta, \quad (41)$$

$$-1 \leq \sin \theta \leq 1 \quad (42)$$

and
$$d = a(1 - e^2) \quad (43)$$

where a is the semi major axis.

8) Therefore:

$$V_{\infty}(\text{Newton}) = -\frac{L}{m} \frac{d}{d\theta} \left(\frac{1}{r} \right)$$

$$= \frac{\epsilon L \sin \theta}{dm}$$

— (44)

$$= \frac{\epsilon L \sin \theta}{a(1-\epsilon^2)m} \rightarrow 0$$

because:

$$a \xrightarrow[r \rightarrow \infty]{} \infty \quad - (45)$$

and

$$-1 \leq \sin \theta \leq 1, \quad - (46)$$

a result that confirms that the Newtonian velocity vanishes at a finite r , as shown in another way in UFT 236.